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# The Inertia Torque of the Hooke Joint 

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## History

Already described by Philo of Byzantium in the III century B.C.

Gerolamo Cardano in the XVI century proposed it as motive power transmission joint.

Robert Hooke analysed it in his Helioscopes in the XVII century


Nowadays it is called in several ways: Cardan Joint, Hooke Joint, or as Hooke himself called it, Universal Joint

## Common Applications

## Vehicle Steering Systems <br> 

Thansmission Line

Heany Duty Venicles
Cardan Joint


## Common Applications

## Vehicle Steering Systems

Thansmission Line

Pront/Rear
Driveline

Heany Duty Vehicles


## Common Applications

Vehicle Steeving Systems

## Transmission Line

Front/Rear Duveline

Heany Duty Vehicles


## Common Applications

Vehicle Stering Systems

Thansmission Line

## Friont/Rear Driveline

Hean' Duty Vehicles


## Common Applications



## Heavy Duthy Vehicles

## Schematic

- In order to study the kinematics and the dynamics of the joint, spherical geometry is used
- XYZ is the fixed reference frame
- xyz is the moving reference frame, consistent with the spider



## Nomenclature



- $\boldsymbol{\beta}$ is the angle between the axes of the connected shafts

- $\boldsymbol{\theta}_{4}$ is the angle that $\mathrm{A}_{3}$ makes with the XY plane makes with the XY plane



## Nomenclature

- $\mathbf{I}_{\mathrm{xx}}, \mathbf{I}_{\mathrm{yy}}, \mathbf{I}_{\mathrm{zz}}$ are the mass moments of inertia of the floating link about x , $y$ and $z$ axes
- $\mathbf{M}_{\mathbf{x}}, \mathbf{M}_{\mathbf{y}}, \mathbf{M}_{\mathbf{z}}$ are the torque components acting on the floating link along $\mathrm{x}, \mathrm{y}$ and z axes
- $\boldsymbol{\tau}$ is the angle between YZ-plane and plane $\mathrm{A}_{2} \mathrm{~A}_{3} \mathrm{O}$
- $\varepsilon=1-\left(\mathrm{I}_{\mathrm{yy}} / \mathrm{I}_{\mathrm{zz}}\right)$, usually $\varepsilon=0$
- $\lambda=I_{x x} / 2 I_{z z}$
- $\mathbf{T}$ is the static torque
- $\mathrm{J}=\left(\mathrm{I}_{\mathrm{yy}}+\mathrm{I}_{\mathrm{zz}}-\mathrm{I}_{\mathrm{xx}}\right) / \mathrm{I}_{\mathrm{zz}}=2(1-\lambda)-\varepsilon$



## Nomenclature

- $\mathbf{T}_{\mathbf{1 H}}, \mathbf{T}_{\mathbf{1 V}}$ are the horizontal and vertical rocking torques on input shaft bearings
- $\mathbf{T}_{4 \mathrm{H}}, \mathbf{T}_{4 \mathrm{~V}}$ are the horizontal and vertical rocking torques on output shaft bearings
- $\mathbf{T}_{\mathrm{x}}, \mathbf{T}_{\mathrm{y}}, \mathbf{T}_{\mathrm{z}}$ are the inertia torque components of floating link along $\mathrm{x}, \mathrm{y}$ and z axes
- $\mathbf{T}_{\mathrm{X}}, \mathbf{T}_{\mathrm{Y}}, \mathrm{T}_{\mathrm{Z}}$ are the inertia torque components of floating link along $\mathrm{X}, \mathrm{Y}$ and Z axes
- $\boldsymbol{\omega}$ is the angular velocity of input shaft, assumed constant
- $\omega_{\mathrm{x}}, \omega_{\mathrm{y}}, \omega_{\mathrm{z}}$ are the angular velocity components of floating link along $x-y-z$ axes, respectively


## Kinematics

$A_{2}=\left(0, \cos \theta_{1}, \sin \theta_{1}\right)$
$A_{3}=\left(-\sin \beta \cos \theta_{4},-\cos \beta \cos \theta_{4}, \sin \theta_{4}\right)$
$y$ and $z$ axes must be always perpendicular

$$
A_{2} \cdot A_{3}=0
$$

$A_{2} \cdot A_{3}=-\cos \beta \cos \theta_{1} \cos \theta_{4}+\sin \theta_{1} \sin \theta_{4}=0$

$$
\tan \theta_{1} \tan \theta_{4}=\cos \beta
$$

Differentiating by time it is possible to determine the tranmission ratio:

$$
|\eta|=\frac{\cos \beta\left(1+\tan ^{2} \theta_{1}\right)}{\tan ^{2} \theta_{1}+\cos ^{2} \beta}
$$

## Transmission Ratio



## Inertia Torque Analysis

- It is convenient to express the moment of inertia of the floating link as follow:

$$
\begin{aligned}
& I_{x x}=2 \lambda I \\
& I_{y y}=(1-\epsilon) I \\
& I_{z z}=I
\end{aligned}
$$

Where $\lambda$ and $\varepsilon$ are constants

- To find the inertia torque Euler's moment equations for motion with a fixed point is used, the axes being principal axes:

$$
\begin{aligned}
M_{x} & =I_{x x} \dot{\omega}_{x}+\omega_{y} \omega_{z}\left(I_{z z}-I_{y y}\right) \\
M_{y} & =I_{y y} \dot{\omega}_{y}+\omega_{x} \omega_{z}\left(I_{x x}-I_{z z}\right) \\
M_{z} & =I_{z z} \dot{\omega}_{z}+\omega_{x} \omega_{y}\left(I_{y y}-I_{x x}\right)
\end{aligned}
$$

## Inertia Torque Analysis

- Combining the two sets of equations it is possible to get:

$$
\begin{aligned}
& M_{x}=I \omega^{2} u\left(\theta_{1}\right) \\
& M_{y}=I \omega^{2} v\left(\theta_{1}\right) \\
& M_{z}=I \omega^{2} w\left(\theta_{1}\right)
\end{aligned}
$$

Where $I$ and $\omega$ are constant and the three functions $u\left(\theta_{1}\right), v\left(\theta_{1}\right), w\left(\theta_{1}\right)$ are the following:

$$
\begin{aligned}
& u\left(\theta_{)}=\frac{2 \lambda \dot{\omega}_{x}+\epsilon \omega_{y} \omega_{z}}{\omega^{2}}\right. \\
& v\left(\theta_{1}\right)=\frac{\dot{\omega}_{y}+(2 \lambda-1) \omega_{x} \omega_{z}-\epsilon \dot{\omega}_{y}}{\omega^{2}} \\
& w\left(\theta_{1}\right)=\frac{\dot{\omega}_{z}-(2 \lambda-1) \omega_{x} \omega_{y}-\epsilon \omega_{x} \omega_{y}}{\omega^{2}}
\end{aligned}
$$

## Inertia Torque Analysis

- The angular velocity components $\omega_{x}$ and $\omega_{y}$ can be easily determined as:

$$
\begin{aligned}
& \omega_{x}=\omega \cos \widehat{A_{1} 0 A_{5}} \\
& \omega_{y}=\omega \cos \widehat{A_{1} 0 A_{3}}
\end{aligned}
$$

Where:

$$
A_{1}=\left(\begin{array}{l}
1 \\
0 \\
0
\end{array}\right) A_{3}=\left(\begin{array}{c}
-\sin \beta \cos \theta_{4} \\
-\cos \beta \cos \theta_{4} \\
\sin \theta_{4}
\end{array}\right)
$$

And $\mathrm{A}_{5}$ can be evaluated as:

$$
A_{5}=A_{3} \times A_{2}
$$



## Inertia Torque Analysis

- This gives:

$$
A_{5}=\left(\begin{array}{c}
-\cos \beta \sin \theta_{1} \cos \theta_{4}-\cos \theta_{1} \sin \theta_{4} \\
\sin \beta \sin \theta_{1} \cos \theta_{4} \\
-\sin \beta \cos \theta_{1} \cos \theta_{4}
\end{array}\right)
$$

Which in turn yields:

$$
\begin{aligned}
& \cos \widehat{A_{1} 0 A_{5}}=A_{1} \cdot A_{5}=-\cos \beta \sin \theta_{1} \cos \theta_{4}-\cos \theta_{1} \sin \theta_{4} \\
& \cos \widehat{A_{1} 0 A_{3}}=A_{1} \cdot A_{3}=-\sin \beta \cos \theta_{4}
\end{aligned}
$$

And hence:

$$
\begin{aligned}
& \omega_{x}=-\omega\left(\cos \beta \sin \theta_{1} \cos \theta_{4}+\cos \theta_{1} \sin \theta_{4}\right) \\
& \omega_{y}=-\omega \sin \beta \cos \theta_{4}
\end{aligned}
$$

## Inertia Torque Analysis

$\omega_{z}$ can be found considering that:

$$
\omega_{z}=-\dot{\tau}
$$

Where $\sin \tau$ is determined by geometry :

$$
\begin{aligned}
& \sin \tau=-\sin \beta \cos \theta_{4} \\
& \tau=\arcsin \left(-\sin \beta \cos \theta_{4}\right)
\end{aligned}
$$

Differentiating by time:


$$
-\dot{\tau}=\omega_{z}=-\omega \eta \sec \tau \sin \beta \sin \theta_{4}
$$

## Inertia Torque Analysis

- Differentiating by time it is possible to get the expressions of $\omega_{x}, \omega_{y}$, and $\omega_{z}$. After some algebraic passages the following expressions are obtained:

$$
\begin{aligned}
& \dot{\omega}_{x}=-\eta \omega^{2}\left(\cos \theta_{1} \cos \theta_{4}-\cos \beta \sin \theta_{1} \sin \theta_{4}\right) \\
& \dot{\omega}_{y}=\eta \omega^{2} \sin \beta \sin \theta_{4} \\
& \dot{\omega}_{z}=\omega^{2} \sec \tau \sin \beta\left(\eta^{2}\left(\sin ^{2} \beta \sin ^{2} \theta_{4} \sec ^{2} \tau \cos \theta_{4}-1\right)+\frac{\dot{\eta}}{\omega}\right)
\end{aligned}
$$

In fixed-system coordinates, the inertia torque components are given by the equations:

$$
\begin{aligned}
& T_{X}=-M_{x} \cos (x, X)-M_{y} \cos (y, X)-M_{z} \cos (z, X) \\
& T_{Y}=-M_{x} \cos (x, Y)-M_{y} \cos (y, Y)-M_{z} \cos (z, Y) \\
& T_{Z}=-M_{x} \cos (x, Z)-M_{y} \cos (y, Z)-M_{z} \cos (z, Z)
\end{aligned}
$$

## Inertia Torque Analysis

- Where:



## Inertia Torque Analysis



## Inertia Torque Analysis



## Inertia Torque Analysis



## Approximated Equations

An approximated solution can be evaluated doing the following assumptions:

$$
\begin{aligned}
& \sin \beta \simeq \beta \\
& \cos \beta \simeq 1 \\
& \theta_{4} \simeq \theta_{1}
\end{aligned}
$$

These seem to be reasonable since rarely angle $\beta$ exceeds $30^{\circ}$, especially in high-speed and high-load applications, where it is generally much smaller


## Approximated Equations

- With these assumptions the approximate equations for the inertia torques with terms of first, second and third order in $\beta$, are the following, being $J=\left(I_{y y}+I_{z z}-I_{x x}\right) / I_{z z}=2(1-\lambda)-\varepsilon$ :

$$
\begin{aligned}
& \frac{T_{X}}{I \omega^{2}}=(1-J) \beta^{2} \sin 2 \theta_{1} \\
& \frac{T_{Y}}{I \omega^{2}}=-J \beta \sin 2 \theta_{1}+\frac{1}{3} \beta^{3}(2 J-3) \sin 2 \theta_{1}-\frac{1}{2} \beta^{3} J \sin 4 \theta_{1} \\
& \frac{T_{Z}}{I \omega^{2}}=J \beta \cos 2 \theta_{1}-\frac{1}{6} \beta^{3} J \cos 2 \theta_{1}+\frac{1}{2} \beta^{3} J \cos 4 \theta_{1}
\end{aligned}
$$

Notice that all the functions have a period of $\pi$. Furthermore, the vector sum of the first-order torque components of $T_{Y}$ and $T_{Z}$ constitute a single rotating vector having twice input speed and phased $90^{\circ}$ from the plane of the input torque.

## Approximated Equations

$$
\frac{T_{Y}+i T_{Z}}{I \omega^{2}}=J \beta e^{i\left(2 \theta_{1}+\frac{\pi}{2}\right)}+o\left(\beta^{3}\right)
$$



## Approximated Equtions



## Approximated Equations



## Rocking Torque Analysis

- In the absence of friction, a pin joint cannot transmit moments about its axis. Hence, the inertia component $\left(-M_{z}\right)$ is transmitted only to the output shaft, while the component $\left(-M_{y}\right)$ is transmitted only to the input shaft.
- The torque component $(-\mathrm{Mx})$ is trasmitted toi both shafts, but in different proportions, let's say a fraction $\gamma$ to the output and a fraction $(1-\gamma)$ to the input.

- The determination of $\gamma$ is a difficult task and cannot be determined from rigid-body considerations alone.


## Rocking Torque Analysis

- Since the inertia of the output shaft is not considered, an approximate, constant value of $\gamma$ can be determined from the condition that there will be no output torque due to the inertia torques of the floating link :

$$
\gamma M_{x} \cos \widehat{A_{5} 0 A_{4}}+M_{z} \cos \widehat{A_{2} 0 A_{4}}=0
$$

Where in the chosen model $A_{4}$ is the intersection between the output shaft direction and the sphere:

$$
\begin{aligned}
& A_{4}=(-\cos \beta, \sin \beta, 0) \\
& \cos \widehat{A_{5} 0 A_{4}}=A_{5} \cdot A_{4}=\sin \theta_{1} \cos \theta_{4}+\cos \beta \cos \theta_{1} \sin \theta_{4} \\
& \cos \widehat{A_{2} 0 A_{4}}=A_{2} \cdot A_{4}=\cos \theta_{1} \sin \beta
\end{aligned}
$$

## Rocking Torque Analysis

- Introducing the parameter J and using the approximated equations, it is possible to obtain a constant value for $\gamma$, thus not depending on the input angle:

$$
\begin{gathered}
J=\left(I_{y y}+I_{z z}-I_{x x}\right) / I_{z z}=2(1-\lambda)-\epsilon \\
\gamma M_{x} \cos \widehat{A_{5} 0 A_{4}}+M_{z} \cos \widehat{A_{2} 0 A_{4}}=0 \\
\gamma \simeq \frac{-J}{2-J}
\end{gathered}
$$

On the other hand the fraction transmitted to the input is:

$$
1-\gamma=\frac{2}{2-J}
$$

## Rocking Torque Analysis

- The horizontal and vertical rocking torques transmitted to the input shaft can be expressed as:

$$
\begin{aligned}
& T_{1 H}=\vec{M}_{y} \cdot \vec{j}+(1-\gamma) \vec{M}_{x} \cdot \vec{j} \\
& T_{1 H}=-v\left(\theta_{1}\right) \cos \beta \cos \theta_{4}+(1-\gamma) u\left(\theta_{1}\right) \sin \beta \cos \theta_{4} \sin \theta_{1} \\
& T_{1 V}=\vec{M}_{y} \cdot \vec{k}+(1-\gamma) \vec{M}_{x} \cdot \vec{k} \\
& T_{1 V}=v\left(\theta_{1}\right) \sin \theta_{4}-(1-\gamma) u\left(\theta_{1}\right) \sin \beta \cos \theta_{4} \cos \theta_{1}
\end{aligned}
$$

Where $\vec{i}, \vec{j}, \vec{k}$ are the unit vectors along the fixed X-Y-Z axes.
Using the approximate equations for inertia torques the resuls are:

$$
\begin{aligned}
& \frac{T_{1 H}}{I \omega^{2}}=\frac{1}{2} \beta J \sin 2 \theta_{1} \\
& \frac{T_{1 V}}{I \omega^{2}}=-\frac{1}{2} \beta J\left(1+\cos 2 \theta_{1}\right)
\end{aligned}
$$

## Rocking Torque Analysis

- Similarly the rocking torques on the output shaft can be estimated:

$$
\begin{aligned}
& T_{4 H}=\vec{M}_{z} \cdot \vec{t} \cos \beta+\gamma \vec{M}_{x} \cdot \vec{t} \cos \beta \\
& T_{4 H}=w \cos \beta \cos \theta_{1}-\gamma u\left(\theta_{1}\right) \sin \beta \cos \theta_{1} \sin \theta_{4} \\
& T_{4 V}=\vec{M}_{z} \cdot \vec{k}+\gamma \vec{M}_{x} \cdot \vec{k} \\
& T_{1 V}=w\left(\theta_{1}\right) \sin \theta_{1}-\gamma u\left(\theta_{1}\right) \sin \beta \cos \theta_{4} \cos \theta_{1}
\end{aligned}
$$

Where $\vec{t}$ is the vector $\vec{t}=(\sin \beta, \cos \beta, 0)$
Using the approxmate equations for inertia torques the results are:

$$
\begin{aligned}
& \frac{T_{4 H}}{I \omega^{2}}=\frac{1}{2} \beta J \sin 2 \theta_{1} \\
& \frac{T_{4 V}}{I \omega^{2}}=-\frac{1}{2} \beta J\left(1-\cos 2 \theta_{1}\right)
\end{aligned}
$$

## Rocking Torque Analysis



## Rocking Torque Analysis



## Rocking Torque Analysis



## Rocking Torque Analysis



## Rocking Torque Analysis

- The vector sum of the horizontal and vertical components for both input and output shafts yields a steady torque and a rotating component rotating at twice input speed and leding the plane of the input fork of $90^{\circ}$.

$$
\begin{aligned}
T_{1} & =\frac{1}{2} \beta J I \omega^{2} e^{\frac{3}{2} \pi i}+\frac{1}{2} \beta J I \omega^{2} e^{i\left(2 \theta_{1}+\frac{3}{2} \pi\right)} \\
T_{4} & =\frac{1}{2} \beta J I \omega^{2} e^{\frac{3}{2} \pi i}+\frac{1}{2} \beta J I \omega^{2} e^{-i\left(2 \theta_{1}+\frac{3}{2} \pi\right)}
\end{aligned}
$$

The difference between input and output rocking torque is just in the direction of rotation of the varying vector


## Rocking Torque Analysis

- Input Shaft Rocking Torque





## Rocking Torque Analysis

- Output Shaft Rocking Torque





## Rocking Torque Analysis






## Conclusions

- The inertia torques are of even order, the predominant harmonic being of order 2
- The inertia torque component along the direction of the input shaft axis is of order $\beta^{2}$. The other two components are of order $\beta$
- The dominant terms of the inertia torque components, $T_{Y}$ and $T_{Z}$, vanish if the mass distribution were such that $J$ vanishes. This can be achieved by locating the mass of the floating link as close as possible to the plane of the pin joints
- The approximate solutions are precise for common values of $\beta$ and allow a deep insight into the dynamics of the system
- The approximate solutions for the rocking torques provide a steady torque and a rotating component, equal in magnitude for both input and output shaft


## Conclusions

- The magnitude of the statically induced rocking torques on either shaft is approximately $T \beta$, where $T$ denotes the static transmitted torque by the Cardan joint. At critical speed this will be nearly equal to the rocking torques due to the inertia of the floating link; this leads to an estimation of the critical speed:

$$
\omega_{C R}=\sqrt{\frac{2 T}{I J}}
$$

# Thank you for your attention! 

